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## **SENSITIVITIES FOR TAYLOR-TEST MODEL PARAMETERS**

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# SENSITIVITIES FOR TAYLOR-TEST MODEL PARAMETERS

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**Abstract.** The Taylor-Cylinder test has long been used to calibrate equation-of-state and material-strength models. The process consists of impacting a cylinder of material on a rigid anvil and adjusting the model parameters to predict the resultant shape. Here we discuss the use of Automatic Differentiation (AD) of an Eulerian hydrodynamics model to provide sensitivities that are used as a gradient in the parameter fitting process. We apply AD in the forward and adjoint modes. For comparison, the gradient is also determined by finite differences. We find that adjoint methods provide the most efficient use of computational resources when there are 9 or more parameters.

## INTRODUCTION

The purpose of this project has been to provide sensitivities of results from an Eulerian hydrodynamics computer code (hydrocode) (1) for use in design-optimization and uncertainty analyses. We began (2) by applying an equation-based sensitivity technique used successfully in the early eighties that was applied to reactor-safety thermal-hydraulics problems (3,4), which is called Differential Sensitivity Theory (DST) (5,6). The methodology is as follows: the system of partial differential equations (the forward or physical PDEs) is assembled, and differentiated with respect to the model parameters of interest; the adjoint equations are then determined using the inner-product rules of Hilbert spaces (5); and finally, the resulting adjoint PDEs are solved using straightforward numerical operators. The forward-variable solutions when needed for the adjoint solutions are provided by the original computer code that solves the physical (or forward) problem. In the present hydrocode application, acceptable results were obtained for one-material, one-dimensional problems. The DST results were then improved by means of "compatible" finite difference operators (6,7). We have seen, however, that DST techniques do not produce accurate values for sensitivities to all of the parameters of interest and for problems with discontinuities such as a multi-material problem (8). To obtain accurate

sensitivities for arbitrary numerical resolution a more code-based approach was then tried. Results for two-dimensional problems were obtained (9) by applying Automatic Differentiation of FORtran (ADIFOR, version 3.0) (10).

Here we present sensitivities for Taylor-cylinder impact test calculations. In what follows, we describe AD methods in the context of their use for a hydrocode. We then describe the Taylor-test calculations. This is followed by an examination of the results, accuracy, and computer run times for the ADIFOR-generated code. Finally, we outline our plans for future work.

## AUTOMATIC DIFFERENTIATION METHODS FOR A HYDROCODE

Use of a hydrocode for experiment fitting purposes requires information about how some scalar result (or response,  $R$ ) will change when some code parameter ( $\alpha$ ) is changed. This so-called sensitivity,  $(\partial R / \partial \alpha)$ , is the gradient (or Jacobian) that determines the search direction for obtaining an optimum response. Typically, the gradient is obtained by changing parameters one at a time to form a finite-difference (FD) derivative. This method requires  $N+1$  computer runs to determine sensitivities for  $N$  problem parameters.

Both code- and equation-based differential sensitivity methods can be implemented in either the forward or adjoint mode. By forward and

adjoint, we mean the direction through the solution and in time and space in which the derivative values are obtained. The forward mode is more efficient for determining the sensitivity of many responses to one or a few parameters, while the adjoint mode is better suited for sensitivities of one or a few responses with respect to many parameters. Here we apply ADIFOR in both the forward and adjoint modes.

AD tools require several steps to get from the original code to an executable code that produces sensitivities. A precompiler first analyzes the code and modifies it to include code that calculates the derivatives of interest. In the forward mode the logic is straightforward. The required additional storage is simply added to the original code and the derivatives are calculated along with the forward solution. In the adjoint mode for a non-linear hydrocode the forward calculation must first be completed since the information from the forward calculation is needed in the adjoint or reverse calculation. Independent storage and/or recalculation can provide this information. The second step in the adjoint process is thus to determine and set up the required storage. For a large problem a technique called checkpointing is required. This technique consists of dumping the solution at checkpoints as the forward solution is generated. The complete forward solution is stored from the final checkpoint to the final time of the forward calculation. One then calculates the adjoint solution backward from the final state to the last checkpoint. The forward solution is then calculated and stored from the second-to-the-last to the last checkpoint. The adjoint solution is then generated from the last to the second-to-last checkpoint. This process is repeated until the initial time of the forward calculation is reached and the sensitivities are complete.

## TAYLOR-CYLINDER IMPACT CALCULATIONS

In this subsection we obtain sensitivities for a copper Taylor-cylinder impact test (11). In the experiment, a 0.762-cm diameter 2.54-cm long copper cylinder at 146 or 190 m/s strikes a rigid anvil. We use a Mie-Gruneisen equation of state (EOS) (12) and the Steinberg-Cochran-Guinan strength model (13) to represent the copper. As implemented in the code, the pressure  $p$  and yield strength  $Y$  are given by

$$p = p_H + \rho_0 e^{-\frac{1}{2} p_H} \left( \frac{1}{\rho_0} - \frac{1}{\rho} \right) \quad (1a)$$

$$p_H = c_0^2 \rho \left( \frac{\rho}{\rho_0} - 1 \right) \left( 1 - \frac{\rho}{\rho_0} - 1 \right) (s - 1)^2 \quad (1b)$$

$$Y = Y_0 \left( 1 + \alpha \left( \varepsilon_0^p + \varepsilon_0 \right)^\beta \right) [1 + \gamma p] e^{-\frac{\delta}{\rho} \left( \frac{e}{e_m} - e \right)} \quad (2)$$

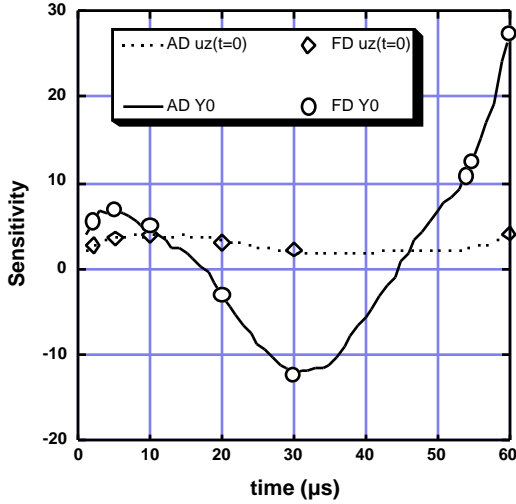
Where  $\rho$  is the density,  $e$  is the internal energy,  $\varepsilon_0$  is the plastic strain, and  $\gamma$  is the pressure-hardening coefficient. The remaining model parameters are defined in Table 1. We intend to use the sensitivities to find the model parameters that best match the final shape of the copper cylinder. The response is therefore chosen to be

$$R = C \left( r_{calc,k} - r_{exp,k} \right)^2. \quad (3)$$

Where  $C$  is a constant,  $r_{exp,k}$  is the  $k^{\text{th}}$  experimental final radius, and  $r_{calc,k}$  is the  $k^{\text{th}}$  calculated final radius. Each of the  $k$  radii is at the same axial location. The sensitivities of this response to the initial conditions and the EOS and strength parameters along with their definitions and nominal values are given in Table 1. If one normalizes by multiplying each sensitivity by its nominal parameter value, the parameters can be ranked as to their importance. We see that the nominal EOS density is the most important followed by the cylinder initial velocity and the strain hardening exponent. The table also compares the AD sensitivities to the FD sensitivities determined with a fractional parameter perturbation size of  $10^{-7}$ . The agreement is only fair for several of the parameters. We compare the AD and FD results by looking at the time-dependence of the sensitivities. Two of the important sensitivities are shown in Fig. 1. The agreement is quite good. Examination of the  $c_0$  sensitivity that did not agree well shows that it agreed well to approximately 30  $\mu\text{s}$ ; then diverged as is shown in Fig. 2. As seen in the figure a smaller perturbation fraction ( $10^{-8}$ ) agrees better at late time. Close examination of all of the FD sensitivities shows that large fractions lose accuracy and small fractions display truncation errors. The comparisons are similar for the 146 m/s case. The FD sensitivities for this problem are thus useful for rough confirmation only. The accuracy required when using the sensitivities as a gradient for an optimization process or uncertainty analysis is yet to be determined.

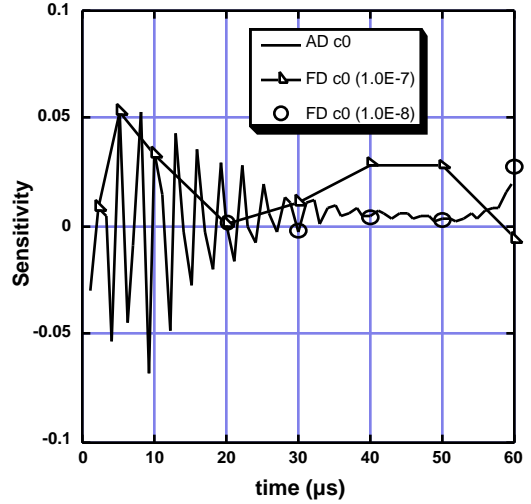
**TABLE 1.** Comparison of the Taylor-cylinder test sensitivities for a 190 m/s impact.

Parameter description and value	AD forward & adjoint sensitivity	AD value normalized	Rank	FD sensitivity (1.0E-7)
Initial density, $\rho(t=0)$ (8.93 g/cm <sup>3</sup> )	0.0038838	3.47E-02	4	0.058777
Initial velocity, $u_z(t=0)$ (0.0190 cm/ $\mu$ s)	3.7952	5.54E-02	2	4.2063
Initial internal energy, $e(t=0)$ (0 Mbar-cm <sup>3</sup> /g)	0.78148	0.00	-	0.99517
Initial stress deviator, $s_{zz}(t=0)$ (0 Mbar)	-0.22942	0.00	-	-0.029852
Shock velocity constant, $s$ (1.489)	-4.6204e-05	6.88E-05	9	-4.4867e-05
Sound speed, $c_0$ (0.3940 cm/ $\mu$ s)	0.022017	8.67E-03	7	-0.0062747
Nominal EOS density, $\rho_0$ (8.93 g/cm <sup>3</sup> )	-0.0078693	7.03E-02	1	-0.062284
Gruneisen ratio, $(2.002)$	-0.00013854	2.77E-04	8	-0.00013871
Linear artificial viscosity constant (0.2)	-7.3006e-05	1.46E-05	11	-0.00010793
Quadratic artificial viscosity constant (2.0)	-1.1642e-06	2.33E-06	13	-1.1569e-06
Strain hardening constant, $\alpha$ (36)	0.00038914	1.40E-02	6	0.00039092
Pre-strain for wrought materials, $\varepsilon_0^p$ (0)	0.029650	0.00	-	0.029319
Strain hardening exponent, $\beta$ (0.45)	0.095916	4.32E-02	3	0.11580
Thermal softening coefficient, $\delta$ (0.001)	-0.0016448	1.64E-06	14	-0.0016989
Melt energy, $e_m$ (0.0571 Mbar-cm <sup>3</sup> /g)	0.00030827	1.76E-05	10	0.00032079
Nominal yield stress, $Y_0$ (0.0012 Mbar)	27.454	3.29E-02	5	27.572
Shear modulus (0.477 Mbar)	-2.7090e-05	1.29E-05	12	-0.0017957



**FIGURE 1.** Comparison of AD- and FD- produced nominal-yield-stress and initial-velocity sensitivities for a 190 m/s impact.

An important issue when choosing whether to use a forward or adjoint method to determine sensitivities is the amount of computer time that each method requires. As mentioned above, the number of parameters versus the number of responses plays a key role in this decision. For this



**FIGURE 2.** Comparison of AD- and FD- produced sound-speed sensitivities for a 190 m/s impact. The FDs were determined with two fractional perturbations ( $10^{-7}$  and  $10^{-8}$ ).

problem we have chosen only one response. The run times for the various methods used to obtain the results in Table 1 are listed in Table 2.

We see that ADIFOR-adjoint uses the least CPU time. For a problem like this one with only one response additional or fewer parameters will not

significantly change the ADIFOR-adjoint run time. ADIFOR-forward and FD runtimes scale linearly with the number of parameters. One could determine the sensitivities for approximately 9 parameters in the adjoint runtime using the ADIFOR-forward or FD methods. This becomes the break-even number for the ADIFOR-adjoint method. In other words, for 9 or more problem parameters the adjoint method will make best use of computational resources. Here we used uniformly spaced checkpoints (every 1  $\mu$ s for the 60 $\mu$ s calculation). This may not be the optimal balance of storage versus recalculation (14). As we gain experience with these methods, minimization of computational time will be better understood and pursued.

**TABLE 2.** SGI Origin 2000 CPU times for Taylor-test calculation with 17 parameters.

Single Forward (CPU sec.)	ADIFOR Adjoint (CPU sec.)	ADIFOR Forward (CPU sec.)	Finite Difference (CPU sec.)
1935	21141	38986	37316

## SUMMARY AND FUTURE WORK

We have applied the automatic differentiation tool ADIFOR (version 3.0) to the MESA2D hydrocode (a Fortran77 code) and have obtained accurate sensitivities for a Taylor-cylinder impact problem in both the forward and adjoint modes. We have determined for problems of this size and duration that the adjoint method is most efficient when there are 9 or more problem parameters. We will apply this capability to experimental-data assimilation and result-uncertainty analysis with this code. We will then extend the capability to parallel codes written in languages other than Fortran77.

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